

IN THE CLAIMS

Please amend the claims as follows:

Claim 1 (currently amended): A ~~division~~ wall of a gas turbine, comprising: 2

1  
a plurality of ~~division~~ wall sections connected in the direction of ~~arrangement of blade~~ 3  
of a set of blades in the gas turbine and ~~forms~~ forming a wall surface having a roughly 4  
substantially circular cross section ~~as a whole~~, the ~~division~~ wall sections being fixed to an 5  
outer end or an inner end of ~~a respective blade of the set of blades in~~ the gas turbine, or being 6  
~~arranged while interposing~~ positioned to interpose a predetermined space between the outer 7  
end of the ~~respective blade to form~~ set of blades and forming a passage wall for high 8  
temperature gas together with a blade surface of the respective blade; and 9

10  
a plurality of gas flow restricting ~~structure which prevents the high temperature gas~~ 10  
~~from passing through a gap~~ devices positioned in gaps formed at ~~a connecting portion~~ 11  
between the ~~division~~ wall sections, respectively, and configured to restrict the high 12  
temperature gas from flowing in a flow direction of the high temperature gas the gaps along 13  
axial and radial directions of the gas turbine ~~from an opening on the upstream side of the high~~  
~~temperature gas in the gap.~~

Claim 2 (currently amended): The ~~division~~ wall according to claim 1, wherein the  
~~blade is~~ set of blades comprises a plurality of stationary ~~blade~~ blades and the ~~division~~ wall is  
a shroud.

Claim 3 (currently amended): The ~~division~~ wall according to claim 1, wherein the  
~~blade is~~ set of blades comprises a plurality of moving ~~blade~~ blades and the ~~division~~ wall is a  
platform.

Claim 4 (currently amended): The ~~division~~ wall according to claim 1, wherein the  
~~blade is~~ set of blades comprises a plurality of moving ~~blade~~ blades and the ~~division~~ wall is a  
~~division~~ ring wall provided in a compartment ~~while~~ and interposing ~~a certain the~~

predetermined space between a tip end of from tip ends of the moving blade blades.

Claim 5 (currently amended): The ~~division~~ wall according to claim 1; wherein the plurality of gas flow restricting structure is devices comprises a plurality of sealing member ~~formed into devices each having a projection shape filling portion configured to fill a~~ respective one of the gap gaps so as to prevent the high temperature gas from leaking outside the passage wall.

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Claim 6 (currently amended): The ~~division~~ wall according to claim 1, wherein the plurality of gas flow restricting structure is comprises a plurality of shielding panel panels ~~which closes the opening positioned to close end openings~~ on the upstream side of the high temperature gas in the gap gaps, respectively.

Claim 7 (canceled)

Claim 8 (currently amended): The ~~division~~ wall according to claim 1, ~~the division wall~~ further comprising a plurality of cooling air blowoff structure for blowing devices ~~configured to blow~~ cooling air into the gap gaps, respectively.

Claim 9 (currently amended): The ~~division~~ wall according to claim 8, wherein the plurality of cooling air blowoff devices comprises a plurality of blowoff opening for blowing ~~the cooling air is openings~~ formed in a side wall ~~surface of the gap~~ surfaces of the gaps, respectively.

Claim 10 (currently amended): The ~~division~~ wall according to claim ~~[[8]]~~ 5, wherein the plurality of cooling air blowoff devices comprises a plurality of blowoff passage for ~~blowing the cooling air is passages~~ formed in the sealing member ~~provided in the gap so as to prevent the high temperature gas from leaking outside the passage wall members,~~ respectively.

Claim 11 (canceled)

Claim 12 (new): A wall of a gas turbine, comprising:

a plurality of wall sections connected to form a wall body having a substantially circular cross section, the wall sections fixing an outer end or an inner end of a set of blades thereon, or being positioned to interpose a predetermined space between the outer end of the set of blades and forming a passage wall for high temperature gas; and

a gas flow restricting device configured to restrict the high temperature gas from flowing in the gaps along axial and radial directions of the gas turbine,

wherein the gas flow restricting device is provided in gaps formed between the wall sections, respectively: